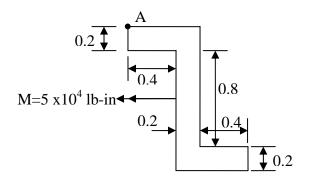
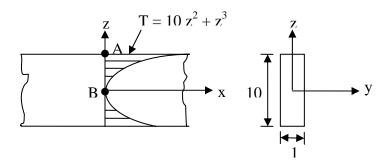
## AE 670 Aerospace Structural Analysis-I Assignment No. 6

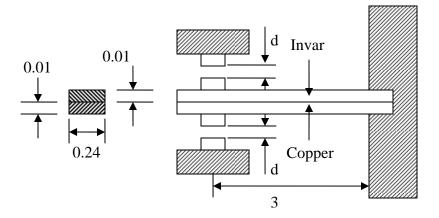
**6.1** Find the bending stress at point A if the Z section is subjected to the bending moment shown.



**6.2** The rectangular beam shown is subjected to a temperature change  $T = 10z^2 + z^3$ . Find the stress at points A and B if  $\alpha = 10^{-5}$  in. / (in.) (°F) and  $E = 10^7$  psi.



**6.3** A bimetallic strip is used in the thermostat, which controls the cabin temperature in a manned space station. The element is of the size and materials shown. Find the clearance d, which should be provided to have the contacts close for a temperature variation of  $\pm 10^{\circ}$ F if the strip is straight at  $72^{\circ}$ F. For invar  $\alpha = 1.1 \times 10^{-6}$  in. / (in.) (°F),  $E = 21 \times 10^{6}$  psi; and for copper  $\alpha = 9.5 \times 10^{-6}$ in. / (in.) (°F),  $E = 17 \times 10^{6}$  psi.



**6.4** The idealized fuselage section shown is subjected to a bending moment of  $10^{-6}$  in –lb about a horizontal axis. The effective areas of the longitudinals and their coordinates are given in the following table. Find the stresses in the longitudinals.

Longitudinal	A <sub>i</sub> in. <sup>2</sup>	y <sub>0</sub> in.	z <sub>0</sub> in.	Z	$\mathcal{L}_0$
1	1.0	0.0	60.0		1 2
2	0.8	14.0	56.0		3
3	0.8	20.0	46.0	•	
4	0.8	24.0	35.0	•	4
5	0.9	26.0	24.0		
6	1.0	27.0	12.0	•	₹5
7	1.0	24.0	4.0		\
8	1.5	15.0	0.0	•	<b>•</b> 6
9	1.0	0.0	0.0		/
					7
					y <sub>0</sub>
					9 8

**6.5** A cantilevered Z-section beam with the cross section given in Prob. 6.1 is 10 in.-lb long and is loaded by a vertical force of 300 lb at the tip. Determine the vertical and horizontal components of the tip deflection.